

PHYS 1131 TEST 1R.

1

(F)

$$(a) \quad \underline{r} = r \cos \theta \hat{i} + r \sin \theta \hat{j}$$

$$= r \cos(\omega t) \hat{i} + r \sin(\omega t) \hat{j} \quad (3)$$

$$(b) \quad \underline{v} = \frac{d\underline{r}}{dt} = r(-\omega) \sin \omega t \hat{i} + r\omega \cos \omega t \hat{j}$$

$$= r\omega (-\sin \omega t \hat{i} + \cos \omega t \hat{j}) \quad (1)$$

$$\underline{a} = \frac{d\underline{v}}{dt} = -r\omega^2 (-\cos \omega t \hat{i} - \sin \omega t \hat{j})$$

$$= -r\omega^2 (\cos \omega t \hat{i} + \sin \omega t \hat{j}) \quad (1)$$

$$= -\omega^2 \underline{r} \quad (1)$$

$$(c) \quad \underline{v} \cdot \underline{r} = r^2 \omega (-\cos \omega t \sin \omega t + \sin \omega t \cos \omega t) = 0 \quad (2)$$

\underline{v} is perpendicular to \underline{r} \quad (1)

$$(d) \quad r = 1.2 \text{ m}$$

$$y - y_0 = 1.8 \text{ m}$$

$$x - x_0 = 9.0 \text{ m}$$

$$y - y_0 = \frac{1}{2} a t^2 \Rightarrow t = \sqrt{2(y - y_0)/a} = \sqrt{2 \times 1.8 / 9.8} = 0.606 \text{ s}$$

$$x - x_0 = v_0 t \Rightarrow v_0 = (x - x_0)/t = 9.0 / 0.606 = 14.9 \text{ ms}^{-1}$$

$$\Rightarrow \text{centripetal acceleration } a_c = v^2/r = (14.9)^2 / 1.2 = 180 \text{ ms}^{-2} \quad (5)$$

(e) Project onto x -axis, for instance:

\Rightarrow from part (b):

$$a = \frac{d^2 x}{dt^2} = -\omega^2 x, \quad x' = r \cos \omega t$$

This is simple harmonic motion.

(3)

2

(17)

$$(a) K = \frac{1}{2}mv^2$$

Force that keeps satellite in circular orbit

$$F = -\frac{GmMe}{r^2} = -m\omega^2 r$$

$$\Rightarrow K = \frac{1}{2}mv^2 = \frac{GmMe}{2r}$$

(3)

$$(b) \frac{1}{2}mv_0^2 - \frac{GmMe}{R_e} = \frac{1}{2}mv^2 - \frac{GmMe}{r}$$

$$= \frac{GmMe}{2r} - \frac{GmMe}{r}$$

$$= -\frac{GmMe}{2r}$$

$$\Rightarrow \frac{1}{2}mv_0^2 = GmMe \left(\frac{1}{R_e} - \frac{1}{2r} \right)$$

$$\Rightarrow v_0 = \sqrt{2GmE_e \left(\frac{1}{R_e} - \frac{1}{2r} \right)}$$

(4)

When $r \rightarrow \infty$, $v_0 \rightarrow \sqrt{2GmE_e/R_e}$ escape speed from Earth.

(1)

$$(c) E = K_A + K_B + K_e + U_{Ae} + U_{Be} + U_{AB}, \quad A, B - \text{satellite}$$

$$\approx K_A + K_B + U_{Ae} + U_{Be}$$

$$= 2(K + M), \quad K_A = K_B = K, \quad U_{Ae} = U_{Be} = U$$

$$= 2 \left(-\frac{GmMe}{2r} \right)$$

$$= -\frac{GmMe}{r}$$

(3)

$$(d) \text{Cons. of momentum } mv - mw = 0 = 2mv$$

\Rightarrow velocity of wreckage immediately following collision $V \geq 0$

$$\Rightarrow E \approx U_{Ae} + U_{Be} = 2U = -2 \frac{GmMe}{r} \quad (4)$$

- (e) Wreckage falls toward Earth. As r decreases, it becomes smaller, and so the kinetic energy K gets larger; wreckage accelerates. Since the angular momentum is zero, wreckage falls directly toward center of Earth (no rotations).

(1b)

3. (a) $mgh = \frac{1}{2}mv_A^2$ energy cons.

$$\Rightarrow N_A = \sqrt{2gh}$$

(2)

(b) $mgh - F_f l = \frac{1}{2}mv_B^2$

$$\Rightarrow mgh - \mu_k m g l = \frac{1}{2}mv_B^2$$

or can use

work-energy theorem

$$\frac{1}{2}mv_B^2 - \frac{1}{2}mv_A^2 = W = -\mu_k mgl$$

$$\Rightarrow v_B = \sqrt{2g(h - \mu_k l)}$$

(3)

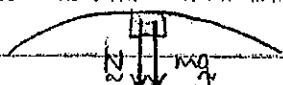
(c) (i) $\frac{1}{2}\mu_k v_B^2 = \frac{1}{2}mv_c^2 + \mu_k g(2r)$

$$\Rightarrow N_c = N_B^2 - 4gr$$

$$= \sqrt{2g(h - \mu_k l - 2r)}$$

(3)

(ii)



(2)

(iii). Forces acting at point C: $\sum F = N + mg = mv_c^2/r$

Block maintains contact with track when $N > 0$

$$\text{i.e. } N = mv_c^2/r - mg > 0$$

$$\Rightarrow \mu_k v_c^2/r > \mu_k g$$

$$\Rightarrow v_c^2 > rg$$

(From part (i)) $\Rightarrow 2g(h - \mu_k l - 2r) > rg$

(iii) cont.

$$\Rightarrow h - \mu_k l - 2r > r/2$$

$$\Rightarrow h > 5r/2 + \mu_k l$$

$$l = 3 \text{ cm}, \quad r = 5 \text{ cm}$$

$$\Rightarrow h > 5l \cdot 5 + 0.3 \cdot 3$$

(4)

$$= 13 \text{ cm}$$

(d) $\frac{1}{2} m v_B^2 = \frac{1}{2} k x^2$

$$v = v_B \Rightarrow m v_B^2 = k x^2$$

$$\Rightarrow k = m v_B^2 / x^2 = m \cdot 2g(h - \mu_k l) / x^2$$

$$= 0.025 \times 2 \times 9.8 (0.18 - 0.3 \times 0.03) / (0.03)^2$$

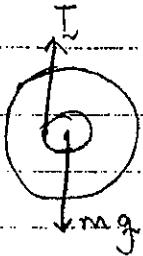
$$= 93 \text{ Nm}^{-1}$$

(2)

(17)

4.

(a)



The tension in the string produces a torque about the centre of mass. (3)

$$(b) \Sigma \tau = R_0 T = I\alpha \quad (= Ia/R_0) \quad (3)$$

$$(c) \Sigma F = mg - T = ma \quad (3)$$

$$(d) (i) mg - T = ma \Rightarrow m \cdot R_0^2 \cdot \frac{T}{I} = mg$$

$$\Rightarrow T(1 + mR_0^2/I) = mg$$

$$\Rightarrow T = \frac{mg}{1 + mR_0^2/I}$$

$$= \frac{mg}{1 + mR_0^2 / (\frac{1}{2}mR^2)}$$

$$= \frac{mg}{1 + \frac{1}{2}R_0^2/R^2} \quad (2)$$

$$(ii) a = (mg - T)/m = g - \frac{g}{1 + \frac{1}{2}R_0^2/R^2}$$

$$= \frac{g(1 + \frac{1}{2}R_0^2/R^2 - 1)}{1 + \frac{1}{2}R_0^2/R^2}$$

$$= \frac{g \cdot \frac{1}{2}R_0^2/R^2}{1 + \frac{1}{2}R_0^2/R^2} \quad (2)$$

(e) Conservation energy

$$\frac{1}{2}I\omega_0^2 = mgl \Rightarrow \omega_0 = \sqrt{2mgl/I}$$

$$\Rightarrow \omega_0 = \sqrt{2mgl/\frac{1}{2}mR^2} = \sqrt{4gl/R^2} = \sqrt{4 \times 9.8 \times 0.8 / (0.03)^2} \\ = 187 \text{ rad.s}^{-1} \quad (4)$$

5.

(a) Initial: $x_{cm} = l + L/2$

Final: $(M+50m)x_{cm} = (d+L/2)M + (d+L)(50m)$.

Centre of mass x_{cm} is same in initial and final case (no ext. forces act).

$$\Rightarrow (M+50m)(l+L/2) = (d+L/2)M + (d+L)50m$$

$$= d(M+50m) + LM/2 + 50mL$$

$$\Rightarrow d = \frac{1}{M+50m} [(M+50m)(l+L/2) - LM/2 - 50mL]$$

$$= \frac{1}{3500+50 \times 75} [(3500+50 \times 75)l - 25mL]$$

$$= \frac{1}{3500+50 \times 75} [(3500+50 \times 75)8 - 25 \times 75 \times 30]$$

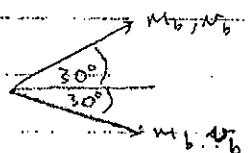
$$= 24 \text{ cm}$$

(6)

(b) speed of ferry relative to the water is zero

(since centre of mass was originally at rest and there were no ext. forces)

(c.)

 $\rightarrow +x$

cons. of momentum

$$0 = 2m_b v_{bx} + (M+50m-2m_b)V$$

$$\Rightarrow V = -\frac{1}{M+50m-2m_b}(2m_b v_{bx} \cos 30^\circ) \text{ toward croc.}$$

$$= \frac{1}{3500+50 \times 75 - 400} (2 \times 200 \times 20 \cos 30^\circ)$$

$$= 0.18 \text{ ms}^{-1} \text{ away from croc.}$$

(6)

(d) Final speed according to observer

$$V' = V + V_{\text{current}} \quad , \quad V_{\text{current}} = 3 \text{ km/h} \\ = 3 \times 10^3 / (60 \times 60) \\ = 0.833 \text{ ms}^{-1}$$

$$\Rightarrow V' = 0.180 + 0.833$$

$$= 1.0 \text{ ms}^{-1}$$

(3)

(1b)

6.

$$(a) I = M/L \int_0^L x^2 dx = M/L \cdot L^3/3 = ML^2/3. \quad (2)$$

$$(b) I_{\text{prop}} = 3 \cdot I = ML^2 = 220 \times 5^2 = 5500 \text{ kg m}^2. \quad (2)$$

$$(c) I_{\text{prop+block}} = I_{\text{prop}} + I_{\text{block}} = ML^2 + mL^2 = 5500 + 25 \times 2^2 \\ = 5600 \text{ kg m}^2. \quad (2)$$

$$(d) \gamma = I\alpha \Rightarrow \alpha = \gamma/I = 7000/5600 = 1.25 \text{ rad s}^{-2}.$$

$$\omega - \omega_0 = \alpha t$$

$$\Rightarrow t = (\omega - \omega_0)/\alpha \quad \omega = 2000 \text{ rpm}$$

$$= 2000 \times 2\pi/60$$

$$= 209.4 \text{ rad s}^{-1}$$

$$\Rightarrow t = 209.4 / 1.25$$

$$= 168 \text{ s.}$$

(4)

(e). Block has travelled a distance

$$s = l\theta \quad , \quad \theta = \theta_0 + \omega_0 t + \frac{1}{2} \alpha t^2.$$

$$= \frac{1}{2} \alpha t^2 = \frac{1}{2} \times 1.25 \times 168^2 = 17640 \text{ rad}$$

$$\Rightarrow s = 2 \times 17640 = 35 \text{ km.}$$

(3)

(f).

$$I_i \omega_i = I_p \omega_p$$

$$\Rightarrow \omega_p = I_i \omega_i / I_p = 5600 \times 209.4 / (5500 + 25 \times 2.5^2)$$

$$= 207 \text{ rad s}^{-1} (= 1980 \text{ rpm})$$

(3)